

AVIATION WEEK

A MCGRAW-HILL PUBLICATION

JAN. 2, 1950



The Grumman MALLARD

The oil industry knows the value of the GRUMMAN MALLARD, as do leaders in such other industries as automobiles, publishing, mining, milling, motion pictures and textiles. The only executive aircraft to combine the speed of a land plane with the versatility of an amphibian, the MALLARD is depended upon by today's corporations for swift transportation of management to places where things happen.



GRUMMAN AIRCRAFT ENGINEERING CORPORATION, BETHPAGE, L. I.

Contractors To The Armed Forces

High Strength

**DOUBLE HEX NUT
CUTS SIZE... WEIGHT OF
AIRFRAME COMPONENTS**



—the famous Self-Locking Red Elastic Collar
protects permanently against IMPACT! VIBRATION!

The new 1974 High Strength Double Hex Nuts have been successfully engineered to accomplish distribution of thread load with minimum weight and scratch diameter. They develop 125,000 psi. min. in max. high strength strength bolts, and are completely interchangeable with existing internal wrenching nuts, yet this unique double hex design saves 60% in weight and 36% in height, compared to the old internal wrenching types. These savings in weight and clearance gain greater importance when multiplied by the additional savings in the size and weight of component parts or fittings which the new design makes possible to employ.

Because of their high strength... extra safety... they fold themselves... and reduce wrenching area, ESNA Type E0



ELASTIC STOP NUTS



OVER 400 TYPES AND SIZES IMMEDIATELY AVAILABLE FROM STOCK

Nuts are being accepted as standard for all bolts over 1/4" in size by many airframe manufacturers. Also, of great importance maintenance men is the fact that no special wrenches are required—any socket wrench will do.

Further—the all Elastic Stop Nuts—the new High Strength Nut reaches self-locking in both fully seated and jammed settings.

SEND A CHALLENGE Send us complete details of your toughest locked nuttable spot. We'll supply test certificate, in experimental quantities. Or, if you want further information, write for literature. Elastic Stop Nut Corporation of America, Union, New Jersey. Representatives and Agents are located in many principal cities.



Research leader
BFG
FIRST IN RUBBER

**"Two-pants suit" idea
saves money and increases safety**

ORDINARY beaded propeller shoes are put out of commission if any part of the shoe is damaged. And the outer section is subject to a lot more wear and tear than the inner section. That's because of the greater abrasive action of the front traveling area. Airframe maintenance men who unpack shoes discovering shoes when the inner section looked good as new.

American Airlines thought there should be a way to make the shoe in two sections, so they brought the problem to B. F. Goodrich. Engineers from both companies studied the problem.

They found that the special BFG method of making beaded rubber equipment made possible a unique way of joining shoes together in two sections.

With the new shoes, if the outer section is damaged, there isn't even a temporary stoppage of the building element in the vital inner shoe section. Replacement is necessary only for the damaged outer section. Tests indicate that the inner section will survive two or even three of the outer pieces, just as a car survives two pairs of pants in a suit's suit. This means a big savings in replacement cost.

And of course the new shoe has all the advantages of BFG beaded rubber: higher efficiency, light weight, simple design, flexibility.

B. F. Goodrich engineers will be glad to discuss the application of the new shoe to the propellers on your planes or drawing boards. Write The B. F. Goodrich Company, Aeronautical Division, Akron, Ohio.

B.F. Goodrich
FIRST IN RUBBER



Another typical aircraft forging by Wyman-Gordon—with an over-all measurement of more than 45" this intricate alloy steel forging is a vital wing support for a modern military bomber.

For applications of such importance the best technique known in forging practice is essential, assuring scale-free surfaces, close dimensional tolerances, uniform minimum weight and maximum strength—There is no substitute for Wyman-Gordon experience.

Standard of the Industry for More Than Sixty Years

WYMAN-GORDON

Forgings of Aluminum, Magnesium, Steel

WORCESTER, MASSACHUSETTS, U. S. A.

HARVEY, ILLINOIS

DETROIT, MICHIGAN

WHO'S WHERE

Changes

► **New Appointments**—Rhele M. Agre, formerly publicity director at National Airlines, has become director of publicity for the Civil Aeronautics Flight Equipment and Repairing Corp. has appointed Gordon McQuinn as assistant manager, with headquarters at the factory in Miami. Dr. William B. Hopkins has been named director of engineering in charge of all engineering and equipment activities for Airborne Accessories Corp.

Mr. Patrick O'Halloran, recently, has joined TWA as a public relations representative in Washington. She resigned three years ago.

Eng. Gen. William T. Thomsen has been advanced to general control of the Department of the Air Force, from his former post as associate general counsel. He succeeds Bradley Shaw former general counsel, who resigned to enter private law practice. Robert T. Grooms, Jr., has been made manager of engineering with the G. M. Corbett Co., Inc., of Pompano, Fla. Dr. R. L. Sullivan, formerly an assistant director for research at NACA, is now assistant director of the Aero Acoustical Lab.

G. J. Randerbush, formerly director of American Airlines, now serves as vice president and supply, a new department in the carrier's organization.

Marie Wartels, assistant director of engineering at the nation's Tulsa overhaul has become assistant vice of the department, in charge of the Tulsa operations. Wausau-Warner Electric Corp. has appointed C. Bruce Weller manager of its Eastern district, with headquarters in New York.

► **Canadian Recruits**—Robert E. Schaefer, formerly with Boeing Aircraft Co. has joined Canadian Ltd., Montreal, as chief technical engineer. William E. O'Neil is now chief of design system. He formerly was with General Electric Co. T. J. Enos, vice president, has joined Pratt & Whitney Co. at Canada and will return his present position as officer and director of Canada.

Elections and Honors

► **New Board Member**—Capt. Ralph S. Bussell, USN (Ret.), has been elected to the board of directors of The Constructor Corp., Memphis, N. Y.

► **Extra Jobs**—Kenneth G. Lockard, vice-president of Capital Aircraft, Inc., has been named a member of the civilian panel which gives advice and preference to the Government of the Army. Appointment was made by Army Committee Lt. Gen. Raymond S. McAllister. Robert Renshaw, STN's executive vice president, has been elected president of the National Capital Panel of the American Public Relations Assn.

INDUSTRY OBSERVER

► **Experimental** use of helicopters by fixed-wing planes, accomplished at Wright Field with Sikorsky and Puma helicopters is expected to be incorporated into the new rescue procedures for larger helicopters to extend range beyond what they could achieve with their own fuel capacity. However, the larger helicopters with auxiliary fuel tanks have range far beyond that of today's operational aircraft. The new Sikorsky HO4S with a normal range of 500 miles, can extend that to 1000 miles with auxiliary tanks, and its structure is designed for towing stations.

► **Northwest Airlines** has run a 1500-hr flight test on engine operations with a Pratt & Whitney R-2800-SD11G coupled to a DC-4 with GAA and PW-W cooperation. Data obtained from the long operation showed it developed better than normal rated power, in various thrust and manifold pressure settings. Test is in line with discussions between GAA and airlines about lengthening the time between overhauls to more than 100 hours. Additional tests are expected before any overhaul intervals are changed.

► **Manufacturers** of small propellers look for more demand for full-fathering propellers to be generated by the new trend toward small twin-engine planes. Improvements in single-engine performance with a built-in "blend-up" is evident as engine quits are quite enough to make the additional overhauls desirable.

► **Problem** of adapting Goodyear cross-section landing wheels to landing gear struts of many of today's lightships has been a factor in delayed acceptance of the wheels. Adapting of the wheels to Cessna gear struts is relatively simple, but Stinson, Luscombe 1A and Aeronca Models 7 and 8 require removal of their struts from the planes to install the cross-section wheels. To overcome this difficulty, propellers of exchange struts have been recommended by Goodyear to its distributors.

► **Cobell Industries Inc.**, Ft. Worth, has sold 10 Grumman TBM torpedo bombers, and three North American SNV trainers to the Yugoslav Army, and is negotiating another contract for sale of 12 Grumman PF fighters. All the planes are war surplus types, manufactured by Cobell and sold to U. S. Navy squadrons before delivery to Yugoslav.

► **New type** wheelbar being introduced by Federal Aircraft Works, Minneapolis, for use in four-place lightships, attach to existing landing gear. Cockpit controls, which hydraulically raise or lower the wheel on the gear, lets the pilot make wheel as his loadings are changed, depend on the on the surface where he is landing.

► **North American Corp.** has a new demand-driven aircraft engine designed to use as an auxiliary engine, and with an automatic control system when the engine supply is exhausted or fails. At 30,000 ft. the engine goes on 10 percent engine.

► **TWA's President** Ralph Denney is warning Lockheed to speed up delivery schedule on the 20 new Constellation ordered for the airline, five of which are now slated for delivery early next spring, and the last for 1951.

► **Synthetic** Lend Lease W. A. Watterton, Greater Aircraft test pilot, has flown to Canada to make fast test flights on the new A. V. Roe Canada Ltd. de Havilland CF-116 night fighter scheduled early in January. Plane has been re-equipped CF-116 by ACAP, receiving Canada dollar 100.

► **Conservation**—British Aircraft Corp. (Australia) will start construction of its first prototype all-weather fighter, which resembles the Lockheed X-40, next summer, following approval by the Australian Minister for Air, and a location of 500,000 Australian pounds to cover cost of design, development and development to prototype stage.

► **British Overseas Airways Corp.** has announced it will not use the Lockheed Constellation that are being converted by the British Ministry of Civil Aviation.

► **A new Finnish** two-seat research plane, the LeDuc 028 developed from the page book, LeDuc 010 is now under construction at Agropoli. The 010 has been launched from pylons on the back of a four engine SF 102 Languette, usually from an altitude of around 17,000 ft. in order to obtain sufficient speed to operate the thermal duct efficiently. The two-place 010 has attained a level speed of 570 mph with only half power at altitude.

The Birdmen's Perch



WHO WILL BE

1950 AEROBATIC CHAMPION?

S. C. Harrison, General Manager of the All American Air Museum, expects only two birds of the current crop to have a shot to compete for the Gulf National Aerobatic Championship Trophy.

This means that last year's top three winners—Steve Howard and Betty Stal—now will lead a qualified field of entrants in clearing the rules they now hold.

The Gulf Open ended contest is open to all—both men and women—on one level with no restriction on the type or make of plane used. Qualifying trials will be held in Miami January 10 to 12. The top ten point scorers will compete for the title and prize in the Masters (January 13-15).

A purse of \$5,750—five splits \$1,750 for 1st place, \$875 for 2nd place, \$575 for 3rd place, and \$375 for 4th place—awaits winners, judging will be held by well-known former aerobatic star.

The way things are shaping up, it looks like the best contest ever. Place no get it. Well, check out this, but better see

with too long to wait to get the contest rules and so called entry list. A note to All American Air Museum, Box 8000, 101 Professional Building, Miami Beach, will bring you complete details by return mail.

HEY...

Do you know that water is wet? That night follows day? That I said I told you so?

You do? Well then, here's another fact equally obvious to brighter pilots: signs and signs remain free longer, time between overhaul periods can be greatly extended, and engine lubrication is more even and more efficient when you use Gulfgrade Aviation Oil—Screen D!



Why Screen D? Because Screen D is an extra oil—Screen D—the world's best designed lubricant and for technically expert experts in the oil industry oil put through Gulf's exclusive Alkylmer process. This makes super expensive products that otherwise are carbon and sludge covered.

LITTLE KNOWN FACTS DEPT.

Overhead: When a better than when making a quick fix. Dollars are lost in a billion's hours and being killed with a spigot.

connector that probably provides you with a finding very whenever and whenever your engine cuts out?

Answer: Obviously, it's a hardware covered, engine-type connector in Perch Pilot.



This (air—well, pilot)—named a Perch Pilot connector (by) for William E. Wood of Haddon, Mass.

A Perch Pilot connector could cover an aircraft for over 250 days on an equal amount of fuel as that required by the X-1 in 250 hours!

See—it's as simple as flying into a connector park in a fog. You offer Perch Pilot potentials with an spending your days in obviously, with your UKF—well, Perch—the Gulf American Department, Gulf Bldg., Pittsburgh 30, Pa.

Gulf Oil Corporation... Gulf Refining Company... marketers of



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AVIATION WEEK

Jan. 2, 1950



TURBOPROP ENGINES, sketched on plans of typical Convaircraft, are power jetties forward and short heavy thrust fans.

Convair Ready to Convert to Turboprop

Could have Liner with new power flying eight months from start.



(This is the second of a series of articles on U. S. jet transport development problems. These articles were prepared by ANTHONY WILSON, Editor Robert H. Wood and News Editor Robert H. Wood, after interviewing executives and engineers of leading Pacific Coast transport manufacturers.)

THE NUCLEAR would have engine an turbine on sides and under stable below

of engineering, and I think Perch, chief engineer of the San Diego division.

►Transport Project—Convair officials make no secret of the fact that the Convair-Liner scheme is well suited for immediate adoption to turboprop power. Accompanying sketch shows how Convair engineers envision the Convair-Liner powered by two Allison T-38 turboprops rated at 2775 square feet shaft horsepower. The T-38 would replace the Pratt & Whitney R-3350 piston engines rated at 2190 hp that are now in service use.

Subfield estimates Convair could have the prototype turboprop Convair-Liner flying within eight months from the time actual work started on such a project. Convair is also trying to suit the U. S. Air Force to a turboprop-powered version of the T-38A navigational trainer adapted from the Convair-Liner scheme.

►Convair Experience—Convair engineers here had probable more experience fitting turboprop engines to air frames than any other U. S. manufacturer. Convair powered the first



CONVAIR TEAM: Oshon, Fink, Schold.

U.S. turboprop installation with its wartime development of the XP81, powered by a General Electric TG-18 turboprop plus a turboshaft. Convair is currently outfitting four Allison XT-40 turboprops (double T-38 units with each pair forming a single set of propellers) in six XP8V-1 turboshaft designs. They long being built for Navy amphibious warfare.

Schold and Fink estimate that it will take another three years to turn out a mass production model turboprop suitable for regular airline operations. They agree that the pace of future transport development will be dictated almost entirely by engine progress.

■ **Engine in Key**—"The power plant is the key to transport development," Schold told Aviation Week. "The airlines we already have are busy to take the power that will be available in the foreseeable future."

Schold pointed out that the turbo prop was the next logical step in transport development, since it could be applied to almost all of the transport designs now in operation and a turbo prop-powered aircraft could use the present airports and traffic control systems.

He also believes that the turbo prop aircraft will have a permanent place in the transport field over medium hauls even after development of turbojet transports. Schold and Fink believe that the turboprop will find it impossible to compete economically with turboprop aircraft over blocks of 500 miles or less.

Schold and Fink believe that the turboprop Convair Lear is offering advantages to airline operators.

■ **More speed**. The turboprop offers more power at low weight and a substantial increase in block speeds over comparable piston engines.

■ **Increased passenger comfort**. The turboprop would eliminate most of the noise and vibration now encountered in piston powered transports.

■ **Scheduled operations**. Some airline operators could take any opera-

tional hours effort in increased stage or increased payloads depending on the requirements of the specific routes. The U.S. airlines in the Convair Lear would need a 2,000 lb. weight saving over piston engines currently used. This is the equivalent of 33 additional passengers or 150 additional gallons of fuel.

Engine interest and ongoing power offered by the turboprop will also permit operation at higher gross weights than are now possible resulting in an additional weight bonus for many payload loads depending on route requirements.

■ **Pressure Required**—Schold and Fink believe that by the time turboprop installation in civil regular airline service USAF technical interest will be strong and for the engine. Since the turboprop studies it has existing efficiency above 20,000 ft. performance will be required for all turboprop aircraft, the engine requires point out. However, some aircraft performance problems will be simplified by the turboprop engine which eliminates any requirement for an additional supercharger by bleeding air from the engine compressor for cabin pressurization.

■ **Financing Problem**—Financing development of turboprop Convair Lear is a major hurdle still to be overcome. If USAF decides to invest in T-1A transport turboprops for high altitude operations, development of the commercial version would be relatively simple financially.

Another possible development, now under way in the making of airlines who now operate Convair Lears to co-operate in the turboprop financing that would provide them with more modern equipment at a fraction of the cost of a new plane.

Oshon noted that that USAF and Navy could aid commercial airline development by financing experimental projects that were desirable to all military manufacturers but which could also be worked commercially.

■ **Oshon Statement**—In regard to

financing new transport prototype development, Oshon goes along with the Aircraft Industries Assn. stand for government financing and for the aircraft industry. However he pointed out that the total equipment requirements of commercial airlines are such that it would be profitable for a manufacturer to supply them if the airlines would help in determining the types of new equipment they wanted.

Convair lost nearly \$10 million on the approximately 200 Convair-Lear it produced as an advanced twin-engine transport type. Oshon said, "Proving that part of this could be obtained to underpinning at the start, it would still have required orders of the magnitude of 400 planes to have broken even. That quantity is within the capacity of the airlines working together on the types of equipment to be purchased."

■ **Evaluating British Lear**—The present British Lear is turboprop and turboprop technology will have some, but not too much effect on long stage competitors with U.S. manufacturers, Oshon believes.

"The British announcement of price and performance tends to show up a general European pattern of current American equipment," he said. "Some of them will apparently be at risk and out to sea what needs out."

"The price that has been asked for these European planes is a structure when viewed in terms of economy devaluation and that is more of a factor. The airlines are not yet ready for turboprops. There are many more problems dealing with airports, economy of operation, loss of time on the ground, and safety in flight, that are all more important than more speed in flight."

In brief terms, looking at Convair's transport future, Oshon says:

"Convair hopes that when better transport plans are built it will beat them. Convair, however, is in business to make profits for its stockholders and therefore does not expect to research financial results for the sake of making some particular type of plane."



NEW NOSE FOR SABRE

North American Sabre fighter jet. The nose of the Sabre, which is faster than most, is made of all-weather lighter resin. Sabre fighter jet, which is faster than most, is made of all-weather lighter resin. Sabre fighter jet, which is faster than most, is made of all-weather lighter resin.

The powerplant's thrust for the Sabre. The D model Sabre is faster than most and slightly heavier than the F100, but more wings and tail section. It is designed as a replacement of the Sabre fighter for short-range high-altitude interception missions. The D model is not to be confused with the all-weather Sabre, which is also developed from the F100, which is also used at Edwards AFB for tests, and which features a new nose along the cockpit and fuselage lines than the original model Sabre.

PAA Holds Off

Carrier will not start Holy Year flights pending CAB decision.

Pan American Airways has agreed to hold off augmentation of its Holy Year charter flights to Rome until the Civil Aeronautics Board decides whether a request to perform duty flights for four months constitutes charter or scheduled service.

The carrier's worldwide schedule came out at hearings in U.S. District Court in New York, where Trans World Airlines filed suit for an injunction (Aviation Week, Dec. 26) which would permanently restrict PAA from flying between the U.S. and Rome. In view of PAA's decision, the court held, no action on the request for the companies.

■ **Not Up To Board**—TWA has filed a brief with CAB asking for dismissal of PAA's proposal to operate the regular charter flights.

Previously, the Board gave PAA permission to operate eight flights a month from January to May and from October to December. PAA then requested permission to operate daily flights in the interim—down from July to September—

and has indicated that it will not fly through with twice-a-week flights if its request to provide daily service is denied.

Both TWA and American Overseas Airlines have complained to CAB against the Board's recent policy that charter flights for PAA and possibly other carriers during 1950. (Shipwreck companies reportedly are considering a protest to the Maritime Commission with regard to refunding air on survivors.)

■ **Agreement**—ADA and CAB's program advances the purpose of the Civil Aeronautics Act and involves schedule changes of the trans-Atlantic route structure. CAB still advised the carrier should limit Atlantic flag service but invert on-again (business) from an irregularly scheduled flights would not be permitted in 1950 but then turned around and favored carrier group charter operations by both scheduled and irregular lines, ADA declared.

TWA's protest was in its own name, LWA's protest was in the name of the carrier. CAB said that it had thousands of requests in 1949-50 during the summer—and expected to increase its capacity in 1950.

Denying the PAA charter, many times in Rome, TWA said the Board's procedure in granting special operating



authority without hearings didn't meet the standards of due play.

"History has established that PAA went into its situation after a long period of planning and eventually gave up its international operations," TWA declared. "The Holy Year charter flights is the most serious case of strategy set down by PAA in its effort to reach its ultimate goal."

■ **PAA's Side**—Pan American charged that TWA's brief was "an attempt by manipulation and false accusations to prevent TWA's scheduled on U.S.-Italy air travel, one of the most profitable of American Airlines, from being scheduled on a regular basis." PAA said that it had thousands of requests in 1949-50 during the summer—and expected to increase its capacity in 1950.



BELL MODEL 45, in passenger version, would carry 12, with cost per seat as low as that for the large fixed-wing planes.

Reveal Details on Transport Helicopters

Bell and Sikorsky get the nod from two New York case applicants.

By Alexander McSweeney

Two of the most active applicants in the New York area helicopter service case before CAB have received an equipment preference for Sikorsky and Bell helicopters, to replace a grounded national fleet of units operating the New York area as far south as Princeton, N. J., and as far north as Rochester, Conn.

In addition to the CAB, the New York Airways, Inc., John J. Scors, Jr., president, proposes to operate three large Bell Model 45 Twin Huey helicopters and six smaller Sikorsky S-52-2 copters.

The Air Commuter, Inc., Robert Dacus, president, proposes to operate seven large Sikorsky S-55 copters if it covers the area with a clockwise route pattern and 11 of the big Sikorskys if an other route relief route pattern is used.

With 10 applicants listed in priced bids, only these two companies have filed evidence supporting their claims and willingness to undertake the service. The Air Commuter volume represents a combination of the company with Metropolitan Airway Corp., under an application now pending before CAB to merge the two organizations.

Meanwhile an editor filed by the



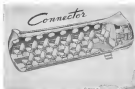
CARGO MAIL VERSION of Bell Model 45 Twin Huey

Helicopter Council of AIA with the CAB gives the first detailed description of performance and specifications of the large Bell and Sikorsky commercial helicopters plus similar data on the Paveco and McDonnell helicopters and on the British Copter Air Horse, all in the same general size and power category.

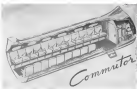
Cost Analysis-Helicopter Council

analysis of purchase costs of three transport-class helicopters, estimated to be from \$120,000 to \$150,000 per unit, shows that they compare favorably with cost per seat of early fixed-wing transport aircraft. The Sikorsky S-55, Douglas DC-3 Lockheed Constellation and Boeing Stearman.

Performance and Specifications—Comparison data on the three large Amer-



PIASECKI HRP-2 offers a 35-passenger "Connector" version and a 21-passenger "Commuter" version, to cruise at 304 mph.



McDONNELL 65C would carry a crew of two and two passengers in this six-engine version.



WITH CARGO compartment, the S-55 would carry crew of two and six passengers.

ican helicopters now in production for military versions and the fourth (McDonnell) ready for production as an order, together with similar information supplied by Cessna on the Air Horse W-11, now being, and a still larger model in design stage the W-11T, are as follows:

• **Bell Model 45 Twin Huey.** Powered with 600 hp Wright R-1320 engine, carries maximum of 12 persons (31 passengers, 1 pilot). Capable of 280 mi flight with 15 min fuel reserve, with 14 passengers and baggage (1900 lb) plus 307 lb reserve and cargo, 114 gal fuel, pilot and necessary equipment. Cruising speed 90 mph at 75 percent power, high speed 118 mph, maximum rate of climb, 1750 ft/min. Fuel consumption 55 gal/hr, gross weight, 7500

lb, weight empty, 4200 lb, cabin size 9 ft. 3 in. by 7 ft. 3 in. by 4 ft. 3 in.

• **Sikorsky S-55.** Powered with 680 hp Pratt & Whitney R-1340 engine, higher suspended performance, with max payload 900 lb payload. Carries maximum of 30 passengers, crew of 2. Capable of 280 mi flight with 15 min fuel reserve with 5 passengers and baggage (1900 lb) plus 307 lb reserve and cargo, 114 gal fuel, pilot and necessary equipment. Cruising speed 90 mph at 75 percent power, high speed 118 mph, maximum rate of climb, 1750 ft/min. Fuel consumption 55 gal/hr, gross weight, 7500 lb, weight empty, 4200 lb, cabin size 9 ft. 3 in. by 7 ft. 3 in. by 4 ft. 3 in.

• **Piasecki HRP-2 (commuter).** Powered with 600 hp Pratt & Whitney R-1340 engine (installation of R-1320

or R-1320 optional without major structural modification) in three engine will carry 11, 15 or 21 passengers plus crew of 2. Capable of 380 mi flight with 15 min reserve with 11 passengers plus 330 lb baggage, or 100 mi flight with 15 min reserve with 15 passengers and 450 lb baggage, or 75 mi flight with 15 min reserve with 21 passengers and no baggage. Cruising speed, 134 mph, high speed, 175 mph, maximum rate of climb, 1680 ft/min, fuel consumption, net gross, gross weight, 11,500 lb; weight empty, 7000 lb; basic weight, including equipment and fuel, 7970 lb to 8182 lb, cabin size, 29 ft by 6 ft. by 5 ft. 6 in.

• **McDonnell 65C.** Powered with two Wright 700 hp engines. Carries maximum of 12 passengers, plus crew of two. Capable of 450 mi flight (cruise) with 12 passengers and 450 lb baggage, or capable of 200 mi flight with 15 min reserve with 1870 lb of actual and cargo, or 160 mi flight with 15 min reserve with 4227 lb of actual and cargo (space limitations prevent additional passenger load). Cruising speed, 87 mph, high speed, 117 mph, maximum rate of climb, 1500 ft/min; fuel consumption, including equipment and fuel, 7970 lb to 8182 lb, weight empty, 5511 lb; cabin size, 12 ft. by 6 ft. 4 in. by 6 ft. 7 in.

• **Cessna Air Horse W-11.** Powered with Rolls Royce 1630 hp engine. Carries maximum of 12 passengers and crew of two. Capable of 300 mi flight with 15 min reserve with 20 passengers and 487 lb baggage at cruise. Cruising speed, 110 mph, high speed, 148 mph, maximum rate of climb, 1210 ft/min; gross weight, 17,500 lb, weight empty, 12,125 lb, cabin size, 10 ft. by 7 ft. 6 in. by 5 ft. 8 in.; fuel consumption, 443 lb/hr.

• **Cessna Air Horse W-11T.** Design study not in flight stage. Powered with two Rolls Royce 1620 hp engine. Carries maximum of 14 passengers and crew

Cost Analysis

	No. Passengers	Price	Total Investment Per Passenger Seat
10-Passenger Helicopter	10	\$128,000—	\$12,800—
Cessna 441	40	130,000	3,250
Douglas DC-6	60	528,000	8,800
Luckland Constellation	60	1,908,000	31,800
Boeing Stearman	75	1,508,000	20,100

of 2. Capable of 280 mi. flight with 15 min. cruise with 54 passengers and 1020 lb. baggage, or 160 mi. flight with 54 passengers and 1635 lb. baggage and cargo. Cruising speed, 116 mph, high speed, 146 mph, maximum rate of climb, 400 ft./min., gross weight, 25,000 lb., weight empty, 15,460 lb., cabin size, 22 ft. by 5 ft. by 6 ft. 6 in.; fuel consumption, 4.85 lb./hr.

Need for Service—Need for opening of helicopter service in the New York area is pointed out by the Port of New York Authority in an exhibit filed with CAB last week.

Of 12 million residents in the area, approximately 5 million reside outside New York City proper. In 1965 over 175 million persons were transported into New York from commuter areas around the city.

The Authority points out that while it is not contemplated that the full airport service can at this time be any substantial assistance in handling the huge traffic volumes in the area, if only a remote fraction of the traffic it moved were efficiently the service will be worthwhile.

Exhibited by the applicants and the Port Authority permit detailed estimates of time savings and volume possible through helicopter airport service, and benefits from passenger and cargo service.

Contingents generally are an im-

portant helicopter business service to provide relief passengers departing at very New York airport to other parts of the country with a plane at some other New York airport, a skybus commuter service between surrounding rural communities and New York, and an airport to downtown-Manhattan service.

Aircraft Industry Is Among Top Twenty

Aircraft manufacturing industry declined 14.8 million pounds of military and civil airplanes in 1949, according to a joint official estimate of the Office of Domestic Commerce and the Civil Aeronautics Administration. Figures indicate a continuation of trend towards increased output since 1947.

Dollar value of all shipments by airplane, engine and propeller companies in 1949 is estimated at \$1,780 million, an increase of \$250 million over 1948. This output made the aircraft industry between 15-20th among all U. S. industries, well down from its 1948 position of first but considerably improved over its 1950 ranking of 46th.

Military production continued to lead by an increasing margin with 1949 production being 31 percent military by volume weight and 62 percent by estimated dollar value. The remainder of production was made up of about 15%

of transport and 3400 personal aircraft. Reforming the substantial difference in average airplane weight, these transports aggregate 4 million airplane pounds while the personal aircraft represent only 1.3 million airplane pounds. This is the third consecutive year that personal aircraft production has declined, 1949 output was only 10 percent that of 1948.

Department of Commerce estimates continental revenue to aircraft production during 1950, figuring \$1.8 billion in total output, of 35-41 million pounds airplane weight, including 145 civil transports and about 3000 personal aircraft.

Tulley Named Head Of Research Group

Arthur H. Tulley, Jr., has been elected executive director of the Aeronautical Research Foundation, replacing Lynn L. Bellinger, who relinquished the post to devote full time to other responsibilities, including development of the Kippel Bellinger Helicopter.

Tulley, assistant director of research at the Worcester Business School, was formerly with Bellinger at "Personal Aircraft Research at Airports," and formerly Massachusetts director of aeronautics.

Bellinger continues as a trustee of the Aeronautical Research Foundation, a nonprofit organization incorporated in Maryland and Massachusetts Institute of Technology faculty members and prominent business citizens. ARF is formed as the national source for developing community acceptance of personal airplanes has been completed, with analysis of data to be substantial shortly in a final report to the National Advisory Committee for Aeronautics.

First Performance Figures for Comet

First official performance figures for the de Havilland Comet, released after four months of extensive flight and ground tests, indicate the British jet transport has a cruise speed of 490 mph, gross weight of 165,000 lb., and a climb rate of 3540 ft./min. climb.

Company officials say potential range of the 36-passenger craft, with 12,000 lb. payload, is 2645 miles, representing a "cruise length" of 2140 miles, plus 200 miles diversion allowance, against a headwind of 30 mph. By reducing payload to 6000 lb., potential range is 3000 miles.

A 45-passenger version, under consideration for shorter hauls, can carry 18,000 lb. payload 1750 miles, plus 200 miles diversion allowance, against a 30 mph headwind.

Miami Races

Small races, stunters are slated as stars reject course change.

Third annual Continental Motors trophy sprint plane race, and the Cessna-sponsored national aerobatic championship competition will be principal events of the 1950 Miami (Fla.) Air America Air Meet opening Jan. 15 at Opa Locka airport for five days.

An air cruise in a high more than 150 planes are expected to participate will leave Miami early Jan. 16 for a visit to Miami, returning Jan. 18.

Contractual race will use the area (outside fairways) aerobatic closed course in front of the grandstand as in 1949. A proposed to plan course was abandoned after it was found that part of the course crossed high tension wires. Steve Wittman, Glendale, Wis., and his racing partner Bill Bernard, winner of the two previous Continental sprints, are expected to be in the 1950 competition, with Steve Wittman Special.

Continental race will be limited to the 16 fastest qualifying planes. Three elimination heats of eight laps (16 mi.) will be run on Friday, Jan. 13, two final heats of 10 laps, Saturday, and a 15-lap final race of the best eight planes from the semifinals, Sunday.

Beverly Elwood, Charleston, S. C., three times winner of the aerobatic title, is expected to defend his title, while Betty Skelton, Tampa, 1948 women's champion will compete this year in the open national championship contest.

First military exhibition flying under the new Defense Department instructions will include an F-56 Shooting Star team of 1 leads Air National Guard pilots, the Rockets.

Other aviation participants will include a show business event by 1949 Miss Betty Bellinger, A.R. McHugh, a 40-year-old 8-24 speedster, a disc-holding exhibit by Navy AD-1 Skyraiders, accidents by a Marine jet F-101 fighters, and a display of Naval Air Reserve in Grumman TBFs and Chance Vought F4Us.

Preparations are being made at three airports in the Miami area: Tamiami, Sunny South, and Opa Locka, to accommodate 1949 aviation plans expected in January at Miami.

Regular meeting of the Personal Aircraft Council planned to coincide with the air maneuvers has been postponed to February 9 to Washington, just west of the present month's cancellation who are expected of the council are expected to be in Miami for a post-mortem successful sprint.



MIAMI AIR RACE HOPEFUL: a novel project. Disputed subject being tested by...



... Designer Paul M. Schenker (in cockpit) who reports that his plane has 230-mph top speed using barrel Continental driving extremely short to prop-mounted behind tail.



MIDWEST RACER built by McDonnell employees in order to meet, but is made of wood.



AEROCAR DISPLAYS SEAT LINES

M. E. Taylor made inside his novel Aerocar three-seat convertible plane after one undergoing tests in CAA approval. He noted that stability has proved satisfactory.

In initial tests, SLS needed a CAA decision on necessary strength requirements in light of dual purpose use. Craft has 180 hp Franklin Duesenberg-powered piston prop.

AERONAUTICAL ENGINEERING



FLYING MODEL embodying Zimmerman's design V 373, which later evolved into...



XTCU-1 Navy fighter, also built by Chance Vought, is also shown in the article. Both are not flown because of constant difficulties after war.

Case for a Convertible Aircraft

Cruising performance and economy of low-aspect-ratio configuration can approach those of conventional craft.

By Charles H. Zimmerman*

On the basis of considerable study and experimental data from wind tunnel and full-scale flight tests, a new approach is possible to build a low-aspect-ratio convertible aircraft that will approach the conventional plane in cruising speed and economy yet take off and land as a helicopter.

And it is evident that the success of such a design depends, greatly, on good engineering. For the job often attached to cross weight, lack of aerodynamic efficiency, and poor compression are avoided.

► **Convertible** Classified—Convertible.

*Aeronautical Research Engineer, National Advisory Committee for Aeronautics. Views and opinions expressed here are based on independent work of the author and are not those of NACA.

result can be divided into two general classes—those using a helicopter rotor as an aviation device operated like a rotor at high speeds, and those using the same rotor for both sustentation at low speeds and propulsive thrust at high speeds.

In this latter case the rotor rotates through an angle of approximately 90 deg. in going from hovering to high speed forward flight. Two general schemes have been proposed to effect this change in rotor axis angle. In one case, sustentation of the rotor relative to the aircraft is changed, in the other case the entire craft is rotated. We deal here with a convertible aircraft of the latter type which has been the subject of a considerable amount of theoretical and experimental study.

Fig. 1 shows principal features of this

convertible aircraft. It comprises an airfoil of very low aspect ratio (moving also as the fuselage), two large, coaxial rotors, tractor propellers in rotor, and vertical and horizontal tail surfaces.

► **Hovering, takeoff, landing**—For a craft of this type to take off and land as a coplaner, it must have a landing gear capable of supporting it with the thrust axis nearly vertical. The gear also should accommodate upblast type landing and must be capable of extension for good crosswind and high-speed performance.

Trailing edge of the rotor is really a large flap serving a variety of purposes. Its primary function is to produce low longitudinal trim near the ground. Without its deflection, the craft will pitch forward, because of aerodynamic interference effect between the ground and trailing edge when near the ground. Deflection of the flap results a considerable reduction in landing gear length over that otherwise necessary. It can be used for longitudinal trim in forward flight or be used as a spring loaded stabilizing flap permitting further airfoil location than otherwise possible.

When hovering, the convertible airplane is a type of rotary wing craft with stability characteristics similar to those of a helicopter. Models which have been flown as coplaners have been steady stable longitudinally, but are subject to lateral oscillations which can be damped readily by rotor motion. Longitudinal control when hovering is achieved by deflection of the two horizontal fins, called sidesteps, simultaneously, control about the vertical axis by deflection of the sidesteps differentially, and control in roll by deflection of the rotors.

► **Rotor Considerations**—One of the major questions in the design of a convertible aircraft that is to use the rotor for both sustentation at low speeds and thrust at high speeds is the extent to which high speed and static thrust must be compromised as compared with the performance of a coplaner rotor on one hand and a conventional airplane propeller on the other. Ability of a rotor as propeller to produce static thrust is given by

$$T = C_{T0} \rho \pi R^2 V^2$$

where T is static thrust, ρ , ratio of point input to kinetic energy per unit mass of the downwash, C_{T0} , coefficient of displacement, R , horsepower input to rotor, and D , propeller diameter.

This well-known relationship states that to support a given weight with a given horsepower, there is a maximum rotor diameter which is dependent upon the factor T/C_{T0} . This factor will ordinarily

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fall between 0.65 and 0.80 for both helicopter rotors and aircraft propellers at their most efficient blade angle at the tip speed is below Mach 0.5.

Fig. 2 shows a variation of static thrust efficiency with blade angle for a four-blade, conventional aircraft rotor designed for maximum efficiency as a propeller at a forward speed of 160 mph. These data, taken from tests at the full-scale tunnel at NACA Langley Lab, indicate that blade planform and twist distributions satisfactory for high speeds also give good values of η_s for the static thrust condition.

Selection of rotor design characteristics from the standpoint of inherent performance therefore makes it difficult to choosing the largest diameter which can be used without too seriously increasing structural and high-speed clearance restrictions.

Transition—For the aircraft under consideration, transition from hovering to high speed flight involves gradual rotation of the main rotor in pitch from steady 90 deg to the horizontal to the small angle of attack required for the cruise. In support the weight at high forward speed. Rotating in hovering flight involves the reverse process. For this to be accomplished satisfactorily, there should be no sudden changes in power or control settings required to maintain level flight and so pronounced visibility.

The craft must remain controllable and preferably the controls should give motion in the same sense about the same body axis at all speeds.

It is obviously desirable that the power required in the transition range be not greater at any speed than that required for hovering.

Longitudinal trim must be possible at any speed.

The rotor must not flap excessively or be subjected to excessive stresses.

Tuned Tests—The NACA has made wind tunnel tests on a model of a convertible-type aircraft over the range of attitudes and powers of interest in transition from hovering to high speed flight. Results have been published in two reports.^{1,2}

The model differed from the true convertible in that the aircraft is now level was designed primarily for high speed and had insufficient reflex down chord for hovering or for flight at speeds less than 50 mph with the maximum power and maximum flying weight.

Fig. 1 shows an outline three-view sketch of the craft. The measured performance, stability, and control characteristics give indications of the behavior to be expected of the true convertible of the same type.

Performance Estimates—In making performance studies for this type of con-



Fig. 1 Various views showing principal features of convertible rotor combination.



Fig. 2 Static thrust efficiency factor vs. angle of attack.



Fig. 3 Planview aspects of wind tunnel model of convertible-type aircraft.

vertible, it was evident soon after the first conception of the craft in 1935 that an analytical procedure was needed for estimating performance in the transition range. A crude approximation method was developed in '36 and '37 based on these assumptions:

- Effects of twist and nonuniformity in the tip-speed velocity distribution were neglected.
- Propellers were assumed to convert 70 percent of engine power into thrust in a direction of the air passing through the propeller disk, the air passing through the disk being assumed to have the velocity $(u/2 + V \cos \alpha)$, where α is the moment of velocity added and the flight velocity.
- Increment of velocity ΔV was assumed to be parallel to the propeller axis of rotation and to add uniformly to the

flight velocity to give the final tip-speed velocity.

- Forces on inflow were assumed to be the same as if the inflow were as a free stream having the direction and velocity of the tip-speed.
- Wing area lift line was assumed parallel to the propeller rotational axis.

When applied to the aircraft under consideration the discussion, these relationships result:

$$T = 2\pi \frac{r}{4} D^2 \left(V \cos \alpha + \frac{\Delta V}{2} \right) \rho$$

$$H_p = \frac{2\pi \frac{r}{4} D^2 \left(V \cos \alpha + \frac{\Delta V}{2} \right) \rho}{\rho V}$$

where T is total thrust of two propellers, H_p total static horsepower applied to two propellers, ρ efficiency with which the inflow momentum is converted into gain in momentum of the air parallel to propeller axis, and D , propeller diameter.

For unaccelerated level flight:

$$W = T \cos \alpha + R \sin \theta$$

$$0 = T \sin \alpha - R \cos \theta$$

where R is resultant aerodynamic force, θ angle of resultant aerodynamic force to the vertical, $\theta = \alpha + \alpha/2$, $\alpha/2$ term of $\alpha/2$ angle of attack between airfoil chord and propeller rotational axis.

$$\tan \alpha' = \frac{V \sin \alpha}{V \cos \alpha + \frac{\Delta V}{2}}$$

$$V_1 = \frac{V \sin \alpha}{\sin \alpha'}$$

where V_1 is the tip-speed velocity.

$$C_L = \frac{1}{2} \rho V_1^2 S C_{L\alpha}$$

where C_L is resultant lift coefficient on airfoil at angle of attack α' , S , wing area.

For a wing of the aspect ratio under consideration (12.7),

$$C_L = 3.0 \sin \alpha'$$

Fig. 4 shows the power required for level flight as indicated by NACA wind-tunnel tests and as given by the approximate method for an assumed value of η_s of 0.70. As will be seen from the figure, the agreement is better than the crude assumption warrant without any allowance of compensating margins.

Stalling—This method of estimating performance in the transition range gives a rough indication of the effective angle of attack of the airfoil and therefore of its likelihood of stalling.

Fig. 5 shows the estimated effective angle of attack α_e (projected) angle of attack for two typical cases representing the same aircraft with two different rotor diameters. For the larger rotor, the effective angle of attack reaches a maximum value of 34.3 deg for some extreme displacements of the smaller in

to keep the maximum effective angle of attack down to 31 deg.

From this figure, it is obvious that if stalling is to be avoided in a conventional aircraft of the general type, the wheel must be capable of pivoting to great high angles of attack, of the order of 35 to 40 deg without stalling. The graph indicates that less the stand-point of wheel steering is an advantage, as in the case of a nose wheel, and that it is a factor that should be kept in mind in the final selection of rotor diameter.

It is obvious that there is also a lower limit to the rotor diameter below which the clearance will fail to prevent stalling. The question of the nature of the wheel which was tested in the Langley full-scale tunnel was set by the exact nature that it is necessary to use a clearance gauge mounted on either side of the wheel. These rollers naturally were near the lower limit in diameter.

It is probable that the relatively high values of increased power required at speeds around 50 to 60 mph (Fig. 4) were due to partial stalling of the main blades as a result of the coverage of the slipstream. Not enough experimental data are available to establish either the upper or lower limits of the ratio of rotor diameter to span which is permissible at different speeds.

• **Rotor Data-NACA tests on the conventional model of the XV20-1** in closed measurements of the pitching moment in left coefficient at various advance ratios through the transition range. These measurements were made with the rotor blades locked so as to represent rigid rotors and also with them free to flap about hinge axis so as to represent flexible rotors and also with them free to flap about hinge axis so as to represent flexible rotors.



Fig. 4. Increase in required power for level flight



Fig. 5. Relative mean geometrical angle of attack, transition flight speed



Fig. 6. Pitching moment coefficient and lift coefficient for rigid rotor

high angle of attack, as is necessary in any event when near the ground. The rotor is dependent on the lower limit of the angle of attack of the blades by changing the large angle inclination to reduce or eliminate the blade turning motion as compensating the flapping.

This seems to be obviously indicated when comparing Figs. 4 and 7 and is evidenced by the fact that a highly loaded, low powered prototype of this aircraft having blades which flapped but did not flap has been flown at an advance ratio of 41 deg in level flight on numerous occasions.

Although it is apparent that longitudinal control can be achieved satisfactorily in the transition range it is also probable that it will be necessary to accept a certain amount of static longitudinal instability. This is evident in a general way from the consideration that steady lift up elevator will be required at some speed in the transition range where the elevator setting for trim when hovering is not needed. The movements of the deficiency can only be corrected from full scale flight experience.

There are at least two possible means for the lack of ability to trim with the flapping blades. One is to use the large trailing edge flap shown in Fig. 1, as an auxiliary longitudinal control at

The NACA wind tunnel tests did not include measurements of advance and angle effectiveness in the transition range but since both rotor and airframe remained within the slipstream during the transition, it can be presumed that the surface attached state effectiveness is desirable. No accurate measurements were made of the static or rotor stability derivatives, concerning those total and lateral stability.

From the known behavior of the full scale prototype flying at 41 deg in the upper portion of the transition range and the behavior of models in hovering flight, coupled with the assumption of a smooth, undisturbed flow on the main airfoils and the tail surfaces throughout the transition range, it seems reasonable to state that no sudden decrease in lateral stability will occur, but that there will be an increasing tendency toward lateral instability in hovering flight as it approached.

• **Flight in Place**—Primary objective in developing a convertible aircraft is to reduce cruising performance and cross-country appearance and to increase the ability to land and take off at a cost. Any convertible aircraft has to do this will almost certainly fail to compete with existing types of conventional planes and helicopters.

Cruising speed and economy of a convertible aircraft of the type under discussion are governed primarily by three factors:

- Aerodynamic efficiency
- Overall cost and weight
- Rotor performance as a propeller

Last two of these factors are important for all convertibles, or all aircraft for that matter. They are especially important in the lower limit of the angle of attack of the blades by changing the large angle inclination to reduce or eliminate the blade turning motion as compensating the flapping.

This disadvantage can be minimized only by taking advantage of the possibility of reducing parasite drag to a minimum by housing the propeller leading edge, and useful load outside the fuselage as a rotor. It is, therefore, the L/D ratio of the aircraft in the transition range is equal to or better than that of a conventional conventional airplane. If it is not done, the low performance aircraft will not be competitive.

In any aircraft with gross weight and useful load for a given useful load and horsepower are essential to a high cruising performance and economical appearance. It is especially important on a low speed ratio: that the weight is kept to a minimum and the useful load is as high as possible.

The principles underlying the need for aerodynamic cleanliness together with low weight and small size for high cruising performance with good economy are

well known by all aircraft designers. On the other hand, attainment of satisfactory propeller efficiency at high cruising speeds with rotors which must be of adequate diameter to sustain the aircraft in hovering flight presents a design problem worthy of discussion.

It is convenient to present propeller characteristics in plots of $V/\Delta D$ vs C_L and C_D vs C_L , where V forward velocity in ft/sec, Δ revolutions per unit, D , propeller diameter, and C_L (a non-dimensional power coefficient) = $0.618 \times \text{hp} / (\text{hp} / \text{ft}^2) \times (\text{ft}^2 / \text{sec}^2)$, being forward speed, Δ , the blade horsepower applied to propeller, and C_D the propeller rotational speed.

• **Propeller Plot**—A typical plot of propeller characteristics is shown in Fig. 8. It gives summary of cost of coefficient of horsepower, forward speed, propeller diameter, and propeller rpm, the values of C_L and $V/\Delta D$ are fixed.

If a design chart is available for a propeller, blade or solidity, number of blades, blade radial section, blade thickness and blade twist, which are to the rotor under consideration, it then is possible to plot from the plot of $V/\Delta D$ vs C_L , the blade angle required to obtain the horsepower under the stated conditions. From the efficiency curve, as shown by the plot of C_D vs C_L , for the chosen blade angle. This point of propeller selection is well known but a repeated need to establish a basis for the following discussion.

As in a given propeller, it is possible to state a value on the chart of $V/\Delta D$ vs C_L , which is the loss at the $V/\Delta D$ and C_L points which correspond to the maximum efficiency for the propeller (all of the discussion neglects Mach number effects, generally, which are in the tip blade number is less than 0.8).

If the propeller is to give optimum performance at cruising speed, the C_L number should be chosen to give a value of C_L and $V/\Delta D$ which will fall on the curve for maximum efficiency. The diameter so determined will be much smaller than that required for sustentation of the aircraft in hovering flight.

Increasing the propeller diameter without changing the rpm is a decrease in $V/\Delta D$ and the propeller blade angle required is less than that for maximum efficiency. If the propeller rpm is decreased to increase $V/\Delta D$, C_L is also increased and the design point tends to move parallel to the line for maximum efficiency rather than back to it.

In general, some improvement in efficiency can be gained by reducing the rpm, but only to a very small extent. It will be probably more desirable to keep the rpm low, the design point back into the maximum efficiency value.

Design experience indicates that for cruising speeds of less than 250 mph, it is probably not worth while to pro-



Fig. 7. C_L vs C_D for selected rotor



Fig. 8. Typical propeller design chart

vide two speed gearing to increase the efficiency of cruising without sacrifice of efficiency as a producer of static thrust. For cruising speeds above 250 mph, it will probably be desirable to use two speed gearing, and increasing speeds at 400 mph, it is an absolutely necessary to provide two speed gearing.

Efficiency—As is indicated from the foregoing discussion, the provision of a rotor diameter large enough to permit the attainment of sufficient static thrust for hovering results in a decrease in propeller efficiency. This decrease tends to become accentuated very rapidly, especially if there is no provision for two speed gearing, so that the selection of a rotor diameter for a convertible of the type under discussion is a matter for extreme investigation in the design stage.

Industry for the efficiency to fall off with increase in diameter stems from the fact that the propeller blade velocity is becoming excessive. This need may be partially counteracted by using a moderate number of blades, increasing blade width, and blade area along the lowest possible section drag.

Unfortunately, there are structural limitations which must be observed. Also, the manufacturing pointed out earlier in regard to the loss of intensity of the tip section and the resultant increase in effective angle of attack, all the while in the transition flight range must be kept in mind.

• **Performance Considerations**—Due to transition phenomena affecting the performance of a low-speed rotor would having wingtip propellers is the reduction in induced drag which results from the propeller torque effect upon the wingtip vortices. Theoretical considerations indicate that the induced drag will be

$$D_i = \frac{1}{2} \rho V^2 \left(1 - \frac{1}{4} \frac{b^2}{D^2} \right)$$

where D_i is induced drag, V , weight, ρ , propeller torque, b , span, and q , flight dynamic pressure.

Some experimental evidence as to available is not sufficient to prove or disprove the validity of the analysis which led to the above relationship.

The experimental value upon as given in the preceding section for the blade angle of attack in cruising flight. The factor b^2/D^2 is a ratio of large in cruising flight, but can make a significant difference in the range of values. Its neglect in performance calculations with respect to possible values of rate of climb, cruising speed, range, and ceiling.

The factor foregoing as additional means for using the lowest possible propeller rotational speeds for a given design.

In connection with the cruising performance of the subject convertible, there is a second very important consideration in addition to that of attainment of high cruising speeds for the power. This is that the aircraft must be able to maintain level flight at a reasonable altitude in a single engine.

A craft of this type must have two engines for reasonable safety, since the propeller disk loading is far too high for satisfactory descent. This consideration also stems from the selection of the ratio of rotor diameter to span, since that ratio is high, it will be possible to have with a single engine the sustained level flight with a single engine.

Stability and control of the type of aircraft in cruising flight present only one unusual problem. The use of adjustable controls is necessary to maintain the lift effect of the rotor but, at the same time, introduces a new factor which tends to decrease the dynamic stability of the craft.

The unbalancing effect is caused by a force in the plane of the disk which which results from the aerodynamic moment necessary to maintain the same rate of turning of the rotor as the rate of change of the angle of attack of the rotor. This is a force which is not convertible of low-speed landing and low blade weight but at a certain point less for a craft with high wing loading and heavy rotor blades.

• **Primary A Propeller-Options**—In general the feasibility of using the rotor of a convertible is fully substantiated at low speeds and propellers at high speeds have ranged from the belief on our part that it cannot be done at all, to the belief, on the other hand, that a conventional helicopter rotor will be satisfactory.

Actual design experience indicates that both of these opinions are wrong.

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PRODUCTION Poles Develop Their First Copter

With earlier developments thwarted by lack of know-how, craft indicates new source of technical information.

The first successful Polish helicopter has been developed at the Gdynia Institute Laboratory (Chief Aviator Tests). Preliminary details indicate that the two place craft has a semi-rigid suspended rotor head, two wooden blades, and a tail anti-torque rotor. Fuselage construction is wood and metal. Landing gear is tricycle type, and drive is by belt for positioning the tail rotor in event of a faulty landing.

Program Slow—Crushed with the country's design in Zdzislaw Bonczak, departmental chief of the design and experimental sections of the Gdynia Institute.

Polish lack of previous helicopter design experience, skilled technicians and workers, and sufficient literature delayed start in development of the craft until the summer of 1948. Work progressed slowly, until in May 1949, when it was decided to have the prototype ready by July 21, anniversary of the People's (Polish) Committee of National Liberation Manifesto.

Component Construction—The major various components are built up as follows:

Fuselage is composed of two main members, welded steel tube and wood, fabric-covered main cabin containing engine, axi-rotor engine, rotor shaft and head, and landing gear, and the tail boom which is a plywood covered wood monocoque structure.

Main rotor has blades of NACA 4402A airfoil, plywood covered. Anti-torque rotor has been obtained by attaching a steel ring to the length of the blades, and by having ribs loosely tensioned. Construction was a matter of standard dimensions and design from root to tip improving rotor performance at a light loss in anti-torque qualities.

Rotor blades, excluding cyclic pitch and precession dynamic stability, are attached with main blades to two air control coupling rings forming the universal joint, located on the driving shaft.

Transmission is by means of a 4th case, helical shaft, through double planetary gear directly above the vertically mounted engine. Reduction gear ratio is stated to be 1.9. Rotor tip speed is given as 110 fpm. Manual

decoupling of motor and rotor is possible in flight.

Anti-torque rotor has two blades absorbing 70% percent of total power at maximum engine. Blade of rotation may be inclined. Tail rotor controls are located in the upper section of the circular tail boom.

Flight Controls—The craft has automatic hydraulic control of collective pitch, providing control of lift by means of throttle alone. The controls of the new craft can be divided into two groups:

Control includes controlling inclination of thrust vector of rotor. Pushing

control column forward causes forward flight, throttle controls speed of ascent and descent, and rudder bar controls rotation of craft around the vertical axis.

Wheel regulates plane of rotation of tail rotor, regulator allows adjustment of collective pitch, permitting choice of engine rpm and rotor speed, clutch allows disarming and disengaging rotor and tail rotor simultaneously, and there is a means for controlling rotor bar independently of rotational movement.

Specifications—Rotor dia. is 28 ft 11 in., tail rotor dia. 5 ft 10 in., distance between main rotor axis and 17 ft 2 in. Rotor speed 1,200 rpm, tail rotor speed 7 ft, 6 in., disk area 456 sq ft, trend 7 ft 34 in.

Weights are given as follows: Empty 836 lb, useful load 440 lb, gross 1276 lb. Empty loading is 1,925 lb/eq. ft.

The following performance is claimed: Maximum load speed 87 mph., maximum cruise speed 71.5 mph., maximum vertical climb 8.2 fpm, maximum forward ascent 31.5 fpm, range 175 mi.

USAF, Navy Bid Information

As Manual Contract Procurement Division action available to Aviation Week the latest bid results, shown on this page. Requests for further information should be addressed to Contracting Officer AMC Wright-Patterson AFB, Dayton Ohio at MAJOR MCPSONE.

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1044415, The Federal Post Company, Chicago, on a bid of \$10,000.
1044416, The Federal Post Company, Chicago, on a bid of \$10,000.
1044417, The Federal Post Company, Chicago, on a bid of \$10,000.
1044418, The Federal Post Company, Chicago, on a bid of \$10,000.
1044419, The Federal Post Company, Chicago, on a bid of \$10,000.
1044420, The Federal Post Company, Chicago, on a bid of \$10,000.



Tests Potentiometers

Resistance percentage bridge, engineered for testing and calibrating precision potentiometers, is announced by Spectrobes, Inc., Sparks, Md. The Model E-1, N-Y Electric resistance percentage of total potentiometer resistance tapped in at any mechanical setting of wiper arm. Accuracy is guaranteed to be better than 99 percent at all settings from 0 to 100 percent of total resistance.

Interchangeable components modified. When using bridge circuit, matching voltage drop across selected standard resistance against voltage drop across potentiometer under test. Accuracy is not affected by setting of potentiometer being checked or by normal variations in supply voltage. Equipment operates on 110v, a.c. power or low-voltage d.c. source. Precision measurements are provided for internal galvanometer.

resin is used for taking one frame per minute, 35 is for may be installed on one roll of film. Camera uses standard AN-A-6 as Super 8 film magazine, specially designed to eliminate film slip.

Operation may be at temperatures from -50 to 110 F at altitudes up to 10,000 ft. Metal parts are treated to resist corrosion.

Forward is made for single exposure operation, timing and coding lights mark film to facilitate location of any given exposure or any marked frame without consulting external synchronization pulse circuit. A compensated, any number of synchronized cameras may be used in parallel group, direct or indirect, eliminating possibility of timing or exposure loss. 30 ft. film, standard exposure load of 50 ft. film, standard synchronization at accurate to 1 millisecond, and camera is designed for use with radio synchronizing link.



Speeds Production

Power screw drive, introduced by Shideprod Inc., 2501 N. Catalina Ave., Chapeau 18, Ill., features hopper feeding, driving speed of more than one screw per second with smaller types, and automatic lighting lamp.

Practically all screw sizes are sizes ranging from No. 1 x 4 to 4 x 4 in. are accommodated in specially designed hopper and driving specific of machine. Drive operates at 110v, a.c. Power source and compressed air supply line source and compressed air supply line.

Hopper positions each screw properly for breaking down work to bit where screw is firmly held in place until one or two threads are engaged. As desired torque is reached, clutch automatically disengages driving spindle. With release of foot pedal, spindle returns for another unit.

Working at starting of work is desired to be finished, same shaft permits one-to-one change of size. Another mechanism enables operator to adjust run caused by wrong unit.

Representative person regularly can set machine for different size or style in

about 30 sec. With high 240 ft. machine requires 21 x 14 in. base, base can be adapted to conveyor line.



Dial Shows Hardness

For quickly checking hardness of nonferrous metal or plastic parts, Barlow-Cushman Co., Rockford, Ill., offers Imperial hand hardness tester. Device is suitable for spot checking of stock in any position or area where there's room for operator's hand.

Styled to be capable of accurate readings even when operated by inexperienced person, unit has conveniently located dial to indicate external hardness.

For Bowing Jobs

New thermosetting resin composition, Adhesive T-3, offered by Ames along Products Co., Apopka, Fla., Waukegan, Ind., bonds huge variety of materials to each other—metal, glass, ceramics, rigid plastics, wood, rubber, etc. Resin cures resistant to water, solvent and does not shrink at weld area bonding.

No primer coating or any assembly drying is required. Surfaces are coated with adhesive, brought together, and held with pressure sufficient to maintain uniform contact. Resulting bond is reported to be strong, moisture- and solvent-resistant, and permanent.

Adhesive is supplied in two parts, resin composition and liquid activator. Two liquid activators are available, one a fast setting type giving rigid initial set (and a short pot life of 20-30 min.), the other a slower reacting type (with a pot life of 4 to 5 hr.).

Adhesive begins to cure as soon as activator is mixed with resin composition. At room temperature, glue line will develop about 1/2 in. strength overnight on one-to-one mix. At 100 F. or with other 70 F. through developer at about a week. Temperature of 100 F. will cause glue line twice as fast as room temperature.

AIR TRANSPORT

Examiner Backs PAA-AOA Union

But he recommends provision for employee protection, not imposed by CAB in previous merger decisions.

Reduction of Pan American Airways' widely-expected plan to buy American Overseas Airlines' properties and 114 miles of trans-Atlantic routes for \$17,650,000 has come from Civil Aeronautics Board Examiner Thomas L. Weiss.

In a report aimed at quelling unsupported charges from the facts of the case, Weiss found that the crowd PAA-AOA merger agreement is consistent with the public interest and will not create a monopoly restricting competition or jeopardizing another airline.

TWA contends that the deal is designed to further Pan American's "boom and bust" policy and would create a new monopoly which might force TWA out of trans-Atlantic operations (Aircraft News, Dec. 31).

Examiner Thomas L. Weiss, Dec. 31, 1957, said that the deal is designed to further Pan American's "boom and bust" policy and would create a new monopoly which might force TWA out of trans-Atlantic operations (Aircraft News, Dec. 31). Weiss found that the deal is designed to further Pan American's "boom and bust" policy and would create a new monopoly which might force TWA out of trans-Atlantic operations (Aircraft News, Dec. 31).

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PAA contends that flight operations, dispatching and maintenance are regulated by AOA might lose first place through the merger. Other personnel might be discharged.

Pan American proposed a full reverse allowance of 25000 considerably less than a year's pay for cost personnel. But Weiss ruled on CAB in patent cases that the deal would create a monopoly restricting competition. He found that the deal would create a monopoly restricting competition. He found that the deal would create a monopoly restricting competition.

ely and would not make TWA as capable of being an effective competitor.

Further, Weiss declared, "The Civil Aeronautics Board and the President have by their actions shown that there is to be no monopoly in international air transportation. CAB's past record in opposing and preventing monopoly by a single carrier without evidence that the board can and will continue to do so, not just police."

In this case, the examiner observed, "It is difficult to be impressed with charges that approval of PAA's acquisition of AOA's temporary certificate, due to expire July 5, 1957, will result in a monopoly."

Changes Discussed—Weiss attached little importance to testimony opposing Pan American's alleged attempts to achieve monopoly by exclusive tactics in bargaining with other governments by acquiring other airlines, and by eliminating competition through division of authority. He said PAA may not have been given to its completion, but added that an attitude was not unusual. The more serious claim arose about PAA's alleged cut throat threat toward competition have not been substantiated, Weiss indicated.

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approval of the merger would help PAA's chances for a dominant seat and that South happened for such a result.

"My South's judgment in conducting the negotiations may have been bad, and he seems to have been in error as to some of his information, in which case he would answer in his defense and stabilizations. But," Weiss continued, "there is nothing to indicate that South knowingly or willfully has requested nothing."

South Airlines—"It is clear that South intended that American should terminate its interests in trans-Atlantic operations and believing her directors felt the same way, he set about doing so by no monopoly in international air transportation. CAB's past record in opposing and preventing monopoly by a single carrier without evidence that the board can and will continue to do so, not just police."

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two monomata pilots that they were entitled to security credit for time in the armed services.

ALPA avoided the aid of a system board of administrative composed of four members, two each from the company and the union. The board held that the two pilots were not employees of Capital when they entered the armed services and, therefore, their seniority should be established in relation to the day they were reemployed.

► **Costs Through**—The monomata pilots then took their case to the U. S. District Court, which decided it had no right to intervene because "final and binding" decision of the administrative board must not subject to court review.

The Court of Appeals decided otherwise, stating that the costs are available to security association members who have no voice in an administrative board's decision.

The Appeals Court thus reversed the merits of the board's findings. It agreed with the board on the fact point: the pilots were not Capital employees when they entered the armed services, and thus had no seniority protection under the draft law. However, under a provision of the contract covering costs of unemployment, these pilots were found entitled to seniority for all time worked for Capital, including the period before they were released prior to the war.

The all-cargo line, Slick, the Flying Tigers and U. S., are all bordered with heavy losses from freight operations during previous positive years. Yet there are some signs of better times ahead.

Slick used to show small profits on cargo services alone during August, September and October—the best three months following its introduction. The Flying Tigers were also in the black during the summer and fall, but earnings came from maintenance and leasing of aircraft by non-scheduled passenger operators, not from cargo services. ► **Slick New Places**—Both Slick and the Tigers would like to buy more second-hand equipment, such as the Douglas DC-4s. Like other cargo carriers, they also hope for continued reductions in ground-handling costs.

Slick plans to acquire a fleet of eight single-engine Northrop Navajo run amphibians with a 2000 lb payload to provide short-haul service. The Navajos will bring San Diego and other southern California cities within 45 min of Slick's main service terminal in Los Angeles, where there will be connections with the company's transcontinental DC-4s flights.

November was the second best traffic month in Slick's history, with 2,600, 400 freight ton-miles flown. An equally high volume was anticipated for December.

► **Rates Rise**—Air freight charges which were quite steady in 1949 as a result of the Civil Aeronautics Board's maximum rate order, took an erratic course at the close of the year.

Specific and general commodity rates in the north-south run out of the Memphis River are heading down with following CAB approval of new tariffs filed by U. S. Airlines. The tariff action is viewed as transatlantic rates, with American, TWA and United raising specific commodity rates about 10 percent in December and planning a similar hike in general commodity rates (especially on mail packages under 50 lb) in Jan. 1.

Eastern Air Lines strenuously objected to the lowered tariff proposed by U. S., alleging that it is suggested post-war investigation CAB rejected the complaint.

LAL and U. S. was waiting a freight rate war by proposing charges averaging 17 cents a ton-mile when its opposing competitors in the third quarter of 1949 were over 62 cents a ton-mile. U. S. replied that the new rates were designed to guarantee additional traffic and are the same or higher than their continental tariffs.

Following CAB's acceptance of U. S. Airlines' rate reductions, Eastern, National and Delta Air Lines acted to make comparable cuts in some of their specific commodity tariffs.



TOP of the cargo list for the first time American Airlines, shown loading near blades, flew over 12 million ton-miles. Time magazine became no.



AIRFREIGHT increase along United Air Lines to city direction dates

Domestic Freight Traffic (Ton-Miles)

Carrier	First 9 Mos. 1949	First 9 Mos. 1948
American	22,515,000	14,553,000
Slick	15,680,000	14,906,000
Slick	16,940,000	15,945,000
TWA	8,907,000	6,795,000
Flying Tigers	8,295,000	9,008,000
Total	72,347,000	64,207,000

Air Freight Sets New High in '49

AA, UAL, Slick run one-two-three in beating previous mark by one-third. Freight charges become erratic.

Air freight, which hit an new tonnage records every winter, set another peak during 1949.

The 16 passenger-carrying domestic truckline's 1949 freight tonnage is expected to total around 95 million—up a third over 1948. It is estimated that the three long-haul, all-rails carriers which were outdistanced last summer—Slick, Airways, The Flying Tiger Line and U. S. Airlines—headed their 35 143 million ton miles in freight.

In 1949, for the first time in the postwar period, a regular passenger-carrying operator led the industry in freight traffic. During the first nine months of the year, American Airlines flew about 22,515,000 freight ton-miles. United Air Lines was in second spot for the period with 17 million freight ton-miles.

► **Slick Third-Slick**, the independent operator which headed the freight tonnage in 1946, 1947 and 1948, slipped to third place in the first nine months of 1949 when it flew 16,940,000 ton-miles. TWA was 5,908,000 ton-miles and The Flying

Liners Last with 8,295,000 ton-miles as the first three of 1949 rounded out the industry's "big five" freight carriers.

Even with the continued growth in air freight traffic, it is doubtful that any of the scheduled operators are relying heavily on it consistently. Passenger-carrying lines don't separate their freight revenues from other costs.

During the first nine months of 1949, the 16 domestic truckline's took in over a dollar from passenger traffic for every five cents derived from freight. Average freight revenue of 15-20 cents a ton-mile compared with about 70 cents a ton-mile for passenger, 60 cents a ton-mile for carrying passengers at the regular air-service rate (20c and 40 cents a ton-mile for flying coach passengers at four cents a mile fare).

► **Outlook Favorable**—Despite these figures, many industry experts are still to love they can build up a profitable freight business guaranteeing no such revenue as passenger traffic. But they refuse to act a target date for that development.



CARRIERS have been developing markets in their own backyard. Slick Airways, for example, used four trucks to carry a one-plane shipment on its "powered sled," hauling ready-to-wear suits and women's clothing. Some carriers —



IN a pinch, land and ports for other lines, as TWA is doing for Slick. All in all, 1949 has been record year for air freight, which has included even lower shipments, at this one by the Flying Tigers. The peak in month freight in passenger.

Regulars Favored In U. S.-Alaska Bids

In a report arriving at the noon of non-scheduled operations on the U. S.-Alaska route, Civil Aeronautics Board spokesman William F. Conck has recommended that two candidates can be permitted to compete with Northwest Airlines for Seattle-Anchorage traffic.

Pacific Northern Airlines, which now operates scheduled service in Alaska at its base in Juneau, should be authorized to extend its bids to Seattle for use only, the examiner stated. He said that Pan American Airways, which is now authorized to fly from Seattle to Ketchikan, Juneau, Wrangell, Fairbanks and Nome, should be permitted to operate a new link from Juneau to Anchorage for one year.

Nonstop Feds—Conck recommended denial of Alaska Airlines' application to operate from Fairbanks to Gust Fork, Alaska, the Twin Cities and Chicago, and from Anchorage and Kodiak, to Seattle, Portland, San Francisco and Los Angeles.

He also urged rejection of Mt. McKinley Airways' application for exemption to fly between Anchorage and Seattle based at Anchorage. Mt. McKinley has been one of the principal non-scheduled operators on the U. S.-Alaska run.

In Public Interest—The examiner decided that it is CAB's duty to "guard jealously" the routes it has granted to certificated operators, and should make available to those scheduled carriers all traffic that would serve to improve their economic well being, and hence satisfy requirements. He added that if irregular operators should suffer monetary loss when GAA carries out its responsibilities it must be considered a business matter that should have been anticipated when the schedules started their existence.

"Then," Conck declared, "in a very real sense the traffic loaded by irregular operators would be considered to be removed as traffic which should be available to the certificated carrier if the public interest would thereby be served."

ICAO Jobs Open

International Civil Aviation Organization will accept applications until Jan. 15 for four positions now open at its secretariat in Montreal.

Positions and their starting salaries are: technical drafting officer, \$7,500, technical officer (communications), \$7,600, technical officer, manual accounts, \$6,600 and supervisor, index cost, \$7,800. Applications forms may be obtained from: Executive Secretary, Air

Coordinating Commission, no. 6835, Commerce Building, Washington 25, D. C.

SHORTLINES

► **As Transport Aids**—Adopted a \$135,000 budget for fiscal 1959 as its seven annual director's meeting. Figure is equal to actual expenditures in last fiscal 1949, when the association operated at \$30,000 less than its budget. All officers were re-elected at the meeting, which also named Cecil Hewitt, Northwest Airlines president; J. W. Miller, Mid-Central Airlines president; and R. H. Davis, Piedmont Airlines president, to the ATA board of directors.

► **Alaska Airlines**—Has returned a CAB exemption to make five one-way flights from Everett, Wash., to Anchorage, Alaska, carrying 75 crew, pursuant to a contract with the Matanuska Valley Farmers Cooperative Assn.

► **American**—In cooperation with its aid agency has started selling three-day "McKinley Airways" tours to New York City. AAs is offering entire-party leaders an opportunity to travel to Mexico but its tour leaders may be invited to the U. S. only as separate a tour of ten or more students or businessmen to take advantage of the educational group rate, which is 20 percent under regular one-way fares.

► **EOAC**—Will take delivery shortly on the first of its new fleet of 25 four-engine 40-passenger Hindley Page Hyman. They will replace Selectair flying mainly over operations on the South and East African routes.

► **Canadian Pacific**—Reportedly is negotiating for additional six flatbeds for freight cars. The first pair (new one added) are expected late in 1951 or early 1952.

► **Canada Air Service**—CAB has offered to postpone the Alaskan carrier's temporary and rate in view of the company's critical financial condition.

► **Capital**—President J. H. Cassinell, during a Wright memorial day address at Kill Devil Hill, N. C., compared the 91 scheduled airline passenger fatalities in 1949 with the 180 persons killed in hunting accidents in only two months of the year. About 13.5 million airline passengers were carried domestically in 1949. Cassinell said that for the ten years from 1939 to 1948, the scheduled airlines flew 63 million passengers with a total of 641 fatalities—considerably fewer than the number of persons who will be killed in hunting accidents in 1949 alone.

► **Example**—Has raised hourly wages five cents in a new contract with the IAM

► **Nationwide Air Transport Service**—CAB has given the large regular carrier an exemption to carry baggage and industrial laborers between the U. S. and British West Indies until Mar. 3, 1950. The Miami Springs, Fla., company has been handling this type traffic during most of the postwar period.

► **Northwest Airlines**—V. W. V. NRA director, has been asked by CAB to cease being concurrently a director of Metropolitan Aviation Corp. The re-tying relationship has not received board approval.

► **Northwest-Canary**'s new Strato-cruiser layer service on domestic flights between following points: Scotch (Hag & Hag French Route) or Island (Hag & Hag) or Bouchon (Hag & Hag) or Wicker's (Delwood) with seats or groups, 51, Martins or Manhattan (each), 75 cents, salt drinks, 25 cents.

► **Seattle-Tacoma Airport** has officially placed the Twin Cities at NWAA's base for international operations.

► **Seattle Airlines**—The all-expense tour operator has completed a two-hour booking system with Capital Airlines.

► **TWA**—Will submit an application to amend its schedule to include the company's annual service next May 2.

► **United**—Has formally filed with CAB its special five-one-way route traffic for Seattle-Chicago-Seattle and San Angelo-Seattle-DC-4 flights on which cargo will be carried. Service will start Jan. 16 at the Bristol airport. More than 75 percent of nearly 3000 flights operated by UAL in 1948 were under its authorization on time or within 15 minutes of schedule, a 37 percent improvement over 1945.

CAB SCHEDULE

See 1—Starting on International scheduled service. (District 1281 at 41)

See 2—Starting on International scheduled service. (District 1281 at 41)

See 3—Starting on International scheduled service. (District 1281 at 41)

See 4—Starting on International scheduled service. (District 1281 at 41)

See 5—Starting on International scheduled service. (District 1281 at 41)

See 6—Starting on International scheduled service. (District 1281 at 41)

See 7—Starting on International scheduled service. (District 1281 at 41)

See 8—Starting on International scheduled service. (District 1281 at 41)

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Will you do your part today? Get a Chest X-Ray. It may mean your life!



Published in the public interest by

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EDITORIAL

From Luxury to Utility

Stirve in the Wind Department

American Airlines bought 50 modern Douglas DC-7s. Only six out of the 50 were special luxury versions equipped with berths.

Now, American has launched its transcontinental jet coach service and has advertised for bids to convert three of its fleet of 50 to air coaches.

American has decided that all three planes, in its own words, will be selected from the full dress luxury category.

From luxury to utility.

Other major airlines that persist in concentrating on "luxury" for the low cost of utility for the many will stand up with the other airlines in U.S. business history.

Utility is all right in its place. But it alone will never support a transcontinental jet coach passenger department again.

Give the Fixed-Base a Chance

There is real merit in suggestions that scheduled airlines use the services of fixed-base operators in smaller cities.

Joseph Canale, president of E. W. Wiggins Airways of Boston, is the spokesman for those who would bring private jets of the Air Transport Association and the National Aviation Training Association into agreement as an experimental program of cooperation.

Fixed-base operators' ground handling facilities have been stressed so far. But there would seem to be an immediate side difference in certain fixed and cargo ship agreements and in offering airline passengers special charter air service to office points.

The airlines have been accused, partly of ignoring the independent fixed-base operator, while at the same time protesting to CAB and Congress about the excessive pay rolls they must maintain at small stations.

The fixed-base operator is already on the way. His two experiences and know-how. As long as he is able to keep looking the wolf from the door he will have at least a nucleus staff on duty during the daylight hours. He will be glad to hire anyone else he needs. And if any other of business men has learned how to cut costs and operate on a minimum budget, it is the fixed-base operator.

Furthermore, he is a home owned, well established business which can enhance the reputation of the airline he serves.

As experiments in the Midwest by Mid-Continent Airlines seem to indicate, the fixed-base operator can function as a sales agency and a business partner for the airline as well as

his own air service, which will continue to serve the community as it has in the past.

If the airline industry means what they say about wanting to cut station costs and encourage businessmen to build and operate, they should give the local operator a chance to show what his experience and energy can do for scheduled air transportation.

Death and Headlines

In mid-December when he delivered his Wright memorial day address at Kitty Hawk, J. H. Carmichael reported that the year's accident death toll on the domestic scheduled airlines had reached 94 persons among 154 million air travelers, who had flown some 60 and a half billion miles in 1950.

This is no more deaths, the industry will continue to strive, but what will probably always be unattainable perfection, a death-free record. And well it should. For we taxpayers have come to expect highest standards of safety in public transportation.

But it may surprise some of us, as it did the audience of Capital Airlines' president at Kitty Hawk, to learn that there were about 300 persons who were killed in landing accidents alone in two recent months. No great headlines flared this total from coast to coast.

In the past 18 years, from 1933 to 1949, the scheduled airlines have carried 63 million passengers with a total of 617 fatalities. That ten-year figure is less than the number of persons who will be killed in landing accidents this year alone.

The National Safety Council reports there were 15,320 passenger fatalities in auto and air accidents in 1949. Some time this year someone will be the unluckiest passenger to be killed in an auto accident. And someone else, Slim Car, involved pointed out, will be the 30,000,000th person to be injured in a car.

There were more than 55,000 fatal accidents on the highways of the United States last year.

Stories of traffic accidents and home fatalities must light notice in the papers. But when an airline accident occurs, the story runs headlines in the press and radio throughout the country. It's a dramatic story, capturing reader interest, and it's believed that no editor would be serious in leaving such a story without "the papers," this airline predicament and, only a short time after one of his own transports had fallen into the Potomac River with loss of life.

"However, a reader must estimate probabilities on his attitude toward such stories."

Airline is a strenuous business; its public appeal is as great a liability after an accident as it is an asset during safe operations. Perhaps because of that very liability, aviation ultimately will become the safest of all means of transportation.

—ROBERT H. WOOD

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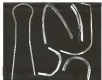
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